NTIA Space record data form

NTIA requires the following data for space related experiments using government shared spectrum. For each transmit frequency, please provide the data for both ends of the transmit-receive link. Use Part A to describe the satellite to ground information. Part B is for all ground to space transmit links.

Part A: Space to Earth Downlink Data

Space to Earth direction is not included

Part B: Ground Stations, Earth to Space link data:

Earth Station Transmitter Data (Required for Each Frequency at Each Earth Station Location)

Earth Station Overview

City (XAL)/ State (XSC)	Lat, Lon	Height [m]	Number of devices
Bellevue/ WA	47.6145, -122.1887	25	Up to 25 (nomadic)
Yakima/ WA	46.6023, -120.5272	335	Up to 25 (nomadic)
Lompoc/ CA	34.6417, -120.4400	36	Up to 25 (nomadic)
San Francisco/ CA	37.7577, -122.4454	140	Up to 25 (nomadic)

Transmit Frequency: 902-928 MHz range (not specified yet)				
State (XSC)	XSC = various			
City Name (XAL)	XAL = various			
Latitude	Lat = various			
(DDMMSS)				
Longitude	Lon = various			
(DDDMMSS)				
Transmit Power	PWR = 1W (max)	TRANSMIT POWER SUPPLIED TO THE ANTENNA		
(PWR)		TRANSMIT POWER UNITS INCLUDE:		
		W = WATT,		
		K = KILOWATT,		
		M = MEGAWATT		
Necessary	1523 kHz	THE WIDTH OF FREQUENCY BAND WHICH IS JUST		
Bandwidth		FORMULAS CAN BE FOUND IN ANNEX LOF THE		
		NTIA MANUAL.		
RF Emissions Data		2-SIDED EMISSION BANDWIDTH VALUES		
-3 dB bandwidth	180 Hz (per Hopping Channel)			
-20 dB bandwidth	550 Hz (per Hopping Channel)			
-40 dB bandwidth	1200 Hz (per Hopping Channel)			
-60 dB bandwidth	1950 Hz (per Hopping Channel)			
Modulation Type	FHSS	THE METHOD USED TO SUPERIMPOSE DATA ON		
		THE CARRIER, EXAMPLE, BPSK, QPSK, GMSK.		
Data Rate	162 b/s or 325 b/s			

Forward Error	Is FEC used? Yes $oxtimes$ No \Box			
Correction Coding	FEC Type: Conv ,			
_	FEC Rate: 1/3 or 2/3 ,			
Total Symbol Rate	488 Hz	DATA RATE COMBINED WITH FEC AND FRAME OVERHEAD RESULTING IN THE TOTAL SYMBOL RATE AT THE INPUTE TO THE SYMBOL MAPPER/MODULATOR.		
Transmit Antenna Polarization (XAP)	XAP = R	POLARIZATIONS INCLUDE: H = HORIZONTAL,		
		V = VERTICAL, S = HORIZONTAL AND VERTICAL, L = LEFT HAND CIRCULAR, R = RIGHT HAND CIRCULAR,		
		T = RIGHT AND LEFT HAND CIRCULAR, J = LINEAR POLARIZATION		
Transmit Antenna Orientation (XAZ)	XAZ = XAZ01 V30	THE EARTH STATION TRANSMITTER ANTENNA MINIMUM OPERATING ANGLE OF ELEVATION (XAZ), V00 TO V90, EXAMPLE, XAZ01 V00		
Transmit Antenna	ANTENNA GAIN2.15 dBi,	EXAMPLE ASSUMING NONGEOSTATIONARY, 16 DBI GAIN, 30 DEGREE BEAMWIDTH, AZIMUTHAL		
Dimensions (XAD)	AZIMUTHAL RANGE_001-360,	RANGE FROM 001-360, SITE ELEVATION OF 357 METERS, AND ANTENNA HEIGHT ABOVE TERRAIN		
	THE SITE ELEVATION ABOVE MEAN SEA	XAD01 16G030B001-360A00357H006		
	THE ANTENNA HEIGHT ABOVE TERRAIN			
	IN METERS0-10,			
	XAD = XAD01 2G120B001-3600357H006			
Transmit Antenna	ANTENNA DIAMETER,			
Additional	ANTENNA EFFICIENCY,			
Information (For				
Antennas)				
Number of	5-may 3 per satellite	NUMBER OF TIMES THE EARTH STATION WILL		
Satellite Contacts	Sinax 3 per satellite	COMMUNICATE WITH THE STATELLITE IN THE		
Supported Per		EARTH TO SPACE DIRECTION (UPINKS) EACH DAY		
Dav				
Expected	300 s	AVERAGE DURATION OF EACH CONTACT		
Duration of Each				
Contact				
Satellite Receive Specifications				
Dessive Antonno				
Receive Antenna Relarization (RAD)	καμ = κ	H = HORIZONTAL,		
FOIdTIZATIOTT (KAP)		V = VERTICAL, S = HORIZONTAL AND VERTICAL		
		L = LEFT HAND CIRCULAR,		
		R = RIGHT HAND CIRCULAR,		
		J = LINEAR POLARIZATION		
Receive Antenna	RAZ = EC	NB= NARROWBEAM		
Orientation (RAZ)		EC = EARTH COVERAGE		

Receive Antenna Dimension (RAD)	ANTENNA GAIN7, BEAMWIDTH70, RAD =	NTIA FORMAT(RAD), EXAMPLE, FOR 16 DBI ANTENNA GAIN AND 30 DEGREE BEAMWIDTH RAD01 16G030B
Type of satellite (State = SPCE) City = Geo or Nongeo	Type = Nongeo	CHOOSE EITHER: GEOSTATIONARY OR NONGEOSTATIONARY
For Geostationary Satellites	Longitude = N/A	IF ANY SATELLITES ARE GEOSTATIONARY, REPORT ITS LATITUDE AS 000000N (XLA AND/OR RLA) AND REPORT ITS LONGITUDE IN DDDMMSS FORMAT (XLG AND/OR RLG).
For Nongeostationary (Orbital Data)	INCLINATION ANGLE, APOGEE IN KILOMETERS, PERIGEE IN KILOMETERS, ORBITAL PERIOD IN HOURSAND FRACTIONS OF HOURS IN DECIMAL, THE NUMBER OF SATELLITES IN THE SYSTEM, ORB,97.7IN00573AP00550PE001.60H01N RT01 ORB,36.6IN00566AP00582PE001.60H02N RT02 ORB,97.6IN00539AP00570PE001.60H03N RT03 ORB,97.5IN00520AP00547PE001.59H04N RT04	IF ANY SATELLITES ARE NONGEOSTATIONARY, REPORT ITS INCLINATION ANGLE, APOGEE IN KILOMETERS, PERIGEE IN KILOMETERS, ORBITAL PERIOD IN HOURS AND FRACTIONS OF HOURS IN DECIMAL, THE NUMBER OF SATELLITES IN THE SYSTEM, THEN T01, EXAMPLE, REM04 *ORB,98.0IN00510AP00510PE001.58H01NRT01, AND FOR SPACE-TO-SPACE COMMUNICATIONS WITH ANOTHER NONGEOSTATIONARY SATELLITE ADD AN ADDITIONAL *ORB FOR IT ENDING IN R01, EXAMPLE, REM05 *ORB,72.9IN03209AP00655PE013.46H01NRR01
For SunSynchronous Nongeostationary Orbits	Mean Local Time of Ascending Node (MLTAN) = T01: 13:23 T02: N/A T03: 10:57 T04: 02:06	MLTAN IS THE ANGLE BETWEEN AN ORBIT'S ASCENDING NODE AND THE MEAN SUN, OFTEN EXPRESSED AS UNIT OF TIME (HH:MM)